NASA/TM-2010-215670



The Lattice and Thermal Radiation Conductivity of Thermal Barrier Coatings: Models and Experiments

Dongming Zhu and Charles M. Spuckler Glenn Research Center, Cleveland, Ohio

NASA STI Program . . . in Profile

Since its founding, NASA has been dedicated to the advancement of aeronautics and space science. The NASA Scientific and Technical Information (STI) program plays a key part in helping NASA maintain this important role.

The NASA STI Program operates under the auspices of the Agency Chief Information Officer. It collects, organizes, provides for archiving, and disseminates NASA's STI. The NASA STI program provides access to the NASA Aeronautics and Space Database and its public interface, the NASA Technical Reports Server, thus providing one of the largest collections of aeronautical and space science STI in the world. Results are published in both non-NASA channels and by NASA in the NASA STI Report Series, which includes the following report types:

- TECHNICAL PUBLICATION. Reports of completed research or a major significant phase of research that present the results of NASA programs and include extensive data or theoretical analysis. Includes compilations of significant scientific and technical data and information deemed to be of continuing reference value. NASA counterpart of peer-reviewed formal professional papers but has less stringent limitations on manuscript length and extent of graphic presentations.
- TECHNICAL MEMORANDUM. Scientific and technical findings that are preliminary or of specialized interest, e.g., quick release reports, working papers, and bibliographies that contain minimal annotation. Does not contain extensive analysis.
- CONTRACTOR REPORT. Scientific and technical findings by NASA-sponsored contractors and grantees.

- CONFERENCE PUBLICATION. Collected papers from scientific and technical conferences, symposia, seminars, or other meetings sponsored or cosponsored by NASA.
- SPECIAL PUBLICATION. Scientific, technical, or historical information from NASA programs, projects, and missions, often concerned with subjects having substantial public interest.
- TECHNICAL TRANSLATION. Englishlanguage translations of foreign scientific and technical material pertinent to NASA's mission.

Specialized services also include creating custom thesauri, building customized databases, organizing and publishing research results.

For more information about the NASA STI program, see the following:

- Access the NASA STI program home page at http://www.sti.nasa.gov
- E-mail your question via the Internet to help@ sti.nasa.gov
- Fax your question to the NASA STI Help Desk at 443–757–5803
- Telephone the NASA STI Help Desk at 443–757–5802
- Write to: NASA Center for AeroSpace Information (CASI) 7115 Standard Drive Hanover, MD 21076–1320

NASA/TM-2010-215670



The Lattice and Thermal Radiation Conductivity of Thermal Barrier Coatings: Models and Experiments

Dongming Zhu and Charles M. Spuckler Glenn Research Center, Cleveland, Ohio

Prepared for the 33rd International Conference and Exposition on Advanced Ceramics and Composites sponsored by the American Ceramic Society Daytona Beach, Florida, January 18–23, 2009

National Aeronautics and Space Administration

Glenn Research Center Cleveland, Ohio 44135

This work was sponsored by the Fundamental Aeronautics Program at the NASA Glenn Research Center.

Level of Review: This material has been technically reviewed by technical management.

Available from

NASA Center for Aerospace Information 7115 Standard Drive Hanover, MD 21076–1320 National Technical Information Service 5301 Shawnee Road Alexandria, VA 22312

The Lattice and Thermal Radiation Conductivity of Thermal Barrier Coatings: Models and Experiments

Dongming Zhu and Charles M. Spuckler National Aeronautics and Space Administration Glenn Research Center Cleveland, Ohio 44135

Abstract

The lattice and radiation conductivity of ZrO_2 - Y_2O_3 thermal barrier coatings was evaluated using a laser heat flux approach. A diffusion model has been established to correlate the coating apparent thermal conductivity to the lattice and radiation conductivity. The radiation conductivity component can be expressed as a function of temperature, coating material scattering, and absorption properties. High temperature scattering and absorption of the coating systems can be also derived based on the testing results using the modeling approach. A comparison has been made for the gray and nongray coating models in the plasma-sprayed thermal barrier coatings. The model prediction is found to have a good agreement with experimental observations.

National Aeronautics and Space Administration

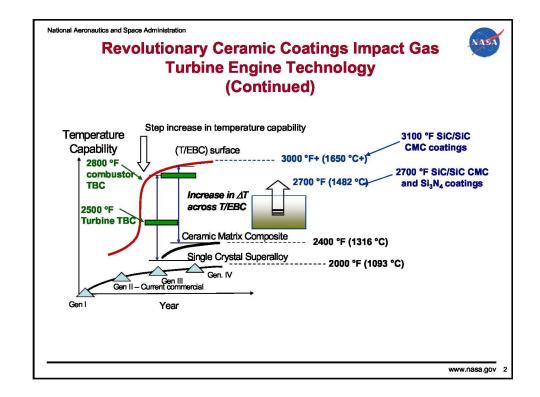
Revolutionary Ceramic Coatings Greatly Impact Gas Turbine Engine Technology

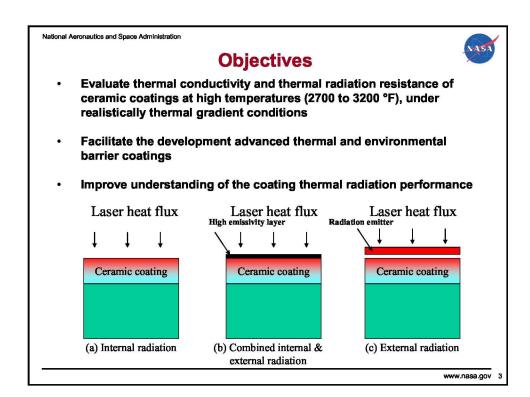


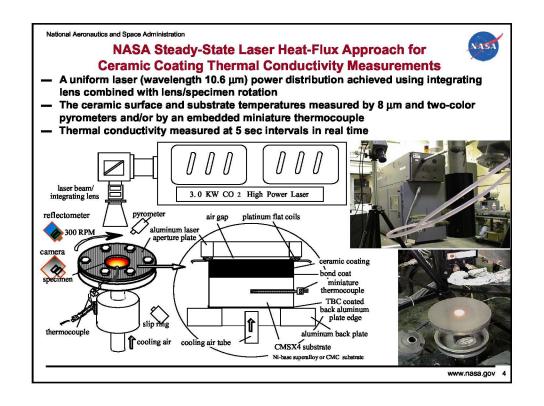
- Ceramic thermal and environmental barrier coating system development goals
 - Meet engine temperature and performance requirements
 - Ensure long-term durability
 - Improve technology readiness
 - Develop design tools and lifing methodologies
- Crucial for envisioned supersonic vehicles: reduced engine emission, improved efficiency and long-term supersonic cruise durability

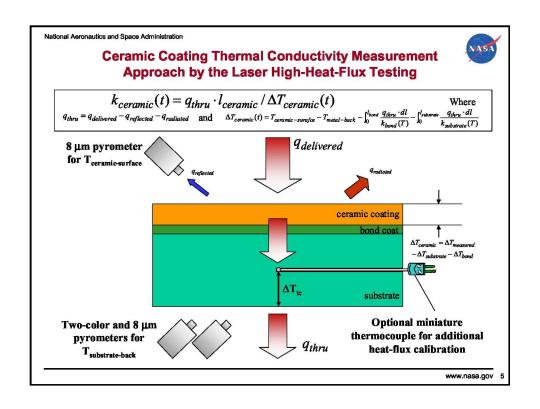


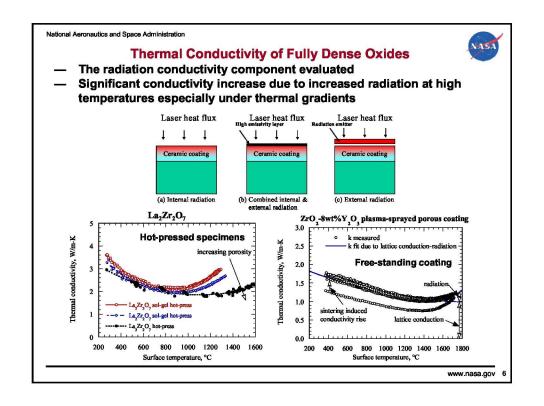
www.nasa.gov

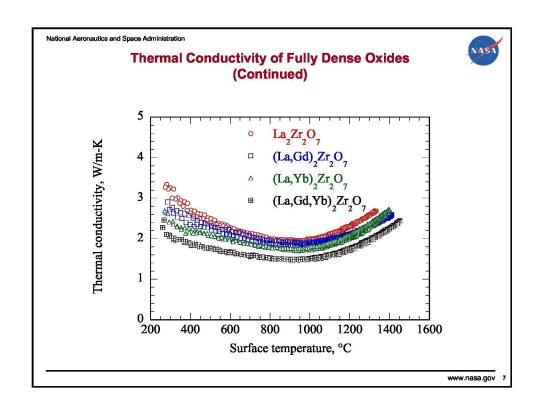


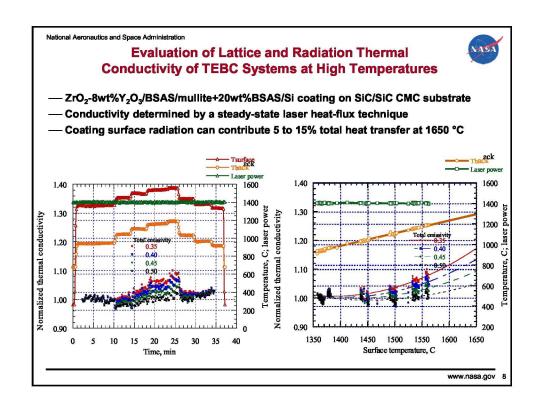












National Aeronautics and Space Administration

Radiative Diffusion Models



- The diffusion conduction equations

$$q_{total} = k_{cond} \frac{dT}{dx} + \frac{16\sigma \cdot n^2 \cdot T_{ave}^3}{3(a + \sigma_s)} \frac{dT}{dx} = \left(k_{cond} + \frac{16\sigma \cdot n^2 \cdot T_{ave}^3}{3(a + \sigma_s)}\right) \frac{dT}{dx}$$

$$k_{effective} = k_{cond} + \frac{16\sigma \cdot n^2 \cdot T_{ave}^3}{3(a + \sigma_s)} = k_{cond} + k_{rad}$$

$$k_{cond} = \text{Intrinsic lattice conductive thermal conductivity}$$

$$k_{cond} = \text{Total heat flux}$$

$$k_{cond} = \text{Intrinsic lattice conductive thermal conductivity}$$

$$k_{cond} = \text{Total heat flux}$$

= effective thermal conductivity

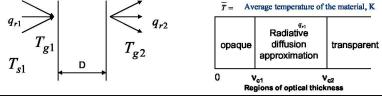
Stefan-Boltzman constant 5.6704x10-8 W/(m2-K4)

Refractive index, 2,2

Absorption coefficient, cm-1

 σ_s = Scattering coefficient, cm⁻¹

 \overline{T} = Average temperature of the material, K



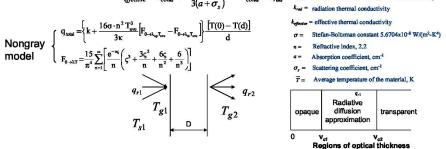
www.nasa.gov

National Aeronautics and Space Administration

Radiative Diffusion Models for Nongray Materials

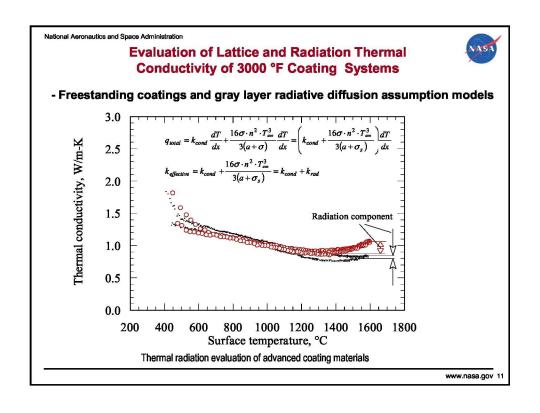


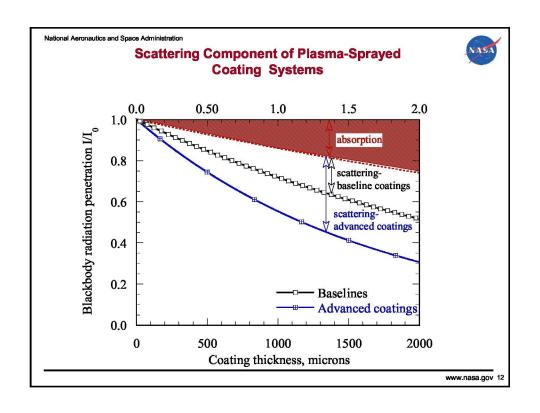
- The diffusion conduction models established for nongray coating materials

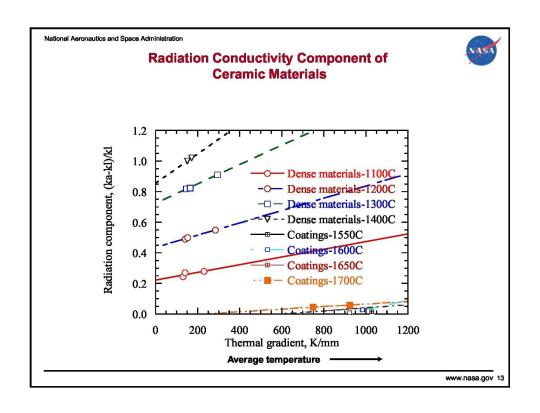


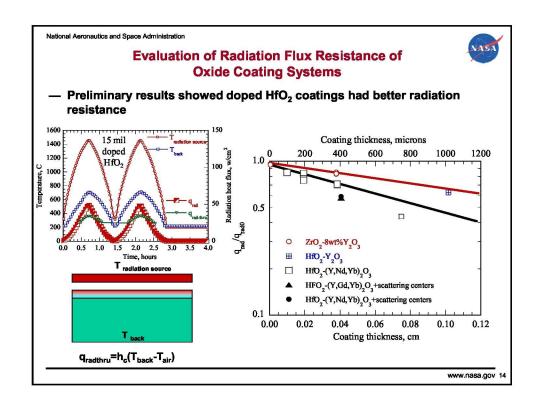
www.nasa.gov 10

NASA/TM-2010-215670









National Aeronautics and Space Administration



Concluding Remarks

- Laser heat-flux approach established for radiation thermal conductivity measurements and advanced coating development
- Lattice and radiation conductivity determined for dense materials and coatings
- The diffusion conduction models established for gray and nongray coating materials
- Scattering and absorption determined for coatings under realistic thermal gradients at high temperatures
- Advanced coatings promising in reducing radiation conductivity

www.nasa.gov 15

REPOR	Form Approved OMB No. 0704-0188				
The public reporting burden for this collection of information is estimated to average 1 hour per response, including the time for reviewing instructions, searching existing data sources, gathering and maintaining the data needed, and completing and reviewing the collection of information. Send comments regarding this burden estimate or any other aspect of this collection of information, including suggestions for reducing this burden, to Department of Defense, Washington Headquarters Services, Directorate for Information Operations and Reports (0704-0188), 1215 Jefferson Davis Highway, Suite 1204, Arlington, VA 22202-4302. Respondents should be aware that notwithstanding any other provision of law, no person shall be subject to any penalty for failing to comply with a collection of information if it does not display a currently valid OMB control number. PLEASE DO NOT RETURN YOUR FORM TO THE ABOVE ADDRESS.					
1. REPORT DATE (DD-MM-YYYY)	2. REPORT TYPE	3. DATES COVERED (From - To)			
01-02-2010	Technical Memorandum				
4. TITLE AND SUBTITLE The Lattice and Thermal Radiation Conductivity of Thermal Barrier Coatings: Models and Experiments		5a. CONTRACT NUMBER			
		5b. GRANT NUMBER			
		5c. PROGRAM ELEMENT NUMBER			
6. AUTHOR(S) Zhu, Dongming; Spuckler, Charles, M.		5d. PROJECT NUMBER			
		5e. TASK NUMBER			
		5f. WORK UNIT NUMBER WBS 984754.02.07.03.16.03			
7. PERFORMING ORGANIZATION NAME(S) AND ADDRESS(ES) National Aeronautics and Space Administration John H. Glenn Research Center at Lewis Field Cleveland, Ohio 44135-3191		8. PERFORMING ORGANIZATION REPORT NUMBER E-17012			
9. SPONSORING/MONITORING AGENCY NAME(S) AND ADDRESS(ES) National Aeronautics and Space Administration Washington, DC 20546-0001		10. SPONSORING/MONITOR'S ACRONYM(S) NASA			
		11. SPONSORING/MONITORING REPORT NUMBER NASA/TM-2010-215670			
12. DISTRIBUTION/AVAILABILITY STA Unclassified-Unlimited Subject Categories: 23, 24, and 27 Available electronically at http://glt This publication is available from the NASA					
13. SUPPLEMENTARY NOTES					
model has been established to corre conductivity component can be expr temperature scattering and absorption	by of ZrO_2 - Y_2O_3 thermal barrier coatings was evaluated us late the coating apparent thermal conductivity to the lattice ressed as a function of temperature, coating material scatter on of the coating systems can be also derived based on the ray and nongray coating models in the plasma-sprayed the h experimental observations.	e and radiation conductivity. The radiation oring, and absorption properties. High testing results using the modeling approach. A			

18. NUMBER OF PAGES

Thermal conductivity; Coatings; Scattering; Absorption; Absorbents; Thermal radiation; Oxides

c. THIS PAGE 17. LIMITATION OF

ABSTRACT

UU

16. SECURITY CLASSIFICATION OF:

a. REPORT

U

b. ABSTRACT

19a. NAME OF RESPONSIBLE PERSON

443-757-5802

STI Help Desk (email:help@sti.nasa.gov)

19b. TELEPHONE NUMBER (include area code)